

# Integrated Simulations using Exascale Multiphysics Ensembles

***Stanford University***  
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Caetano Melone: Verifying Functionality and Performance of HPC Applications with Continuous Integration

Shahab Mirjalili & Henry Collis: Multi-physics Modeling of Laser-induced Ignition in a Multiphase Rocket-module Combustor

Donatella Passiatore: Laser-induced Ignition of a Non-premixed CH<sub>4</sub>/O<sub>2</sub> Mixture in a Rocket-module Combustor





# Verifying Functionality and Performance of HPC Applications with Continuous Integration

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PSAAP III - INTEGRATED SIMULATIONS USING EXASCALE MULTIPHYSICS ENSEMBLES



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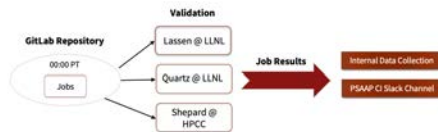
## 1. Motivation and Significance

- HTR-solver: hypersonic aerothermodynamics code and primary software project of the PSAAP-III program at Stanford
- Targeting compatibility on multiple supercomputers; depends on architecture, software dependencies, and other factors
- Our role: improving developer productivity and enabling concurrent developments by automating the verification of changes and identifying performance regressions in real-time

## 3. Implementation



- Utilizing CI tools supported by DOE initiatives: Jacamar Runners through GitLab
- In addition to a commit-level CI, we crafted a suite of jobs that test scale, inter-node communications, and correctness on a scheduled basis
- Automated processes collect performance and functionality data during execution
- Ability to instantly reproduce the environment where a bug occurs helps to resolve issues quickly



## 5. Testing and Verification

### Regression and Correctness Tests

- Testing the merging of features into HTR
- Solver's output vs. analytical solution
- Combination of tests complement each other (problem, numerical scheme, etc.)
- Catch issues early on by scaling to production environment
- Channel Flow, Shock Tube, Taylor-Green, etc.

### Unit Tests

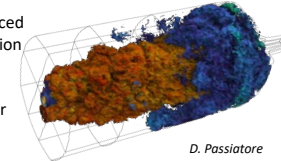
- Small checks for individual modules
- Isolated from the rest of the solver

### Principles

- Changes to code, dependencies, and platforms can have adverse effects on functionality and correctness
- Automated testing can find issues early on with the right amount of code coverage
- Tradeoffs must be made to ensure the verification process is efficient for developers and the CI system

## 2. Background

- Center Objective: Physics simulations of a laser-induced ignition in a rocket combustor, including representation of high-speed flow, turbulence, multiphase flow, combustion, and laser energy deposition
- HTR should be portable, bug-free software; helps our center achieve scientific goals
- Continuous Integration (CI) infrastructure has become commonplace in HPC environments and is leveraged to support this automated framework
- HTR dependencies: Legion, Regent, GASNet, LLVM, and CUDA/HIP

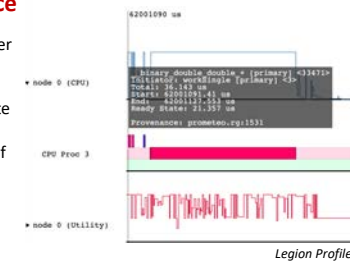


## 4. Measuring Computational Performance

Evaluating performance is a long-term priority of our center

### Resource Utilization and Code Optimization

- Setting a standard for how long it should take to execute certain problems
- This can be done by comparing HTR's results to those of another solver (e.g. S3D)
- Use profiling tools (Legion, NVIDIA Nsight) to identify bottlenecks, areas ripe for optimization, and other aspects where the code can be improved



### Scaling

- When the scalability of an execution is in question, being able to point to data that disproves the influence of certain factors is consequential
  - Ex: thinking a dependency has degraded performance when it is due to other inefficiencies
- We are currently developing tooling that will collect scaling metrics along with covariates
- Can plot trends over time and notify developers when efficiency has suffered

## 6. Conclusions

- CI is an integral aspect of our verification strategy
- We identify issues early so developers can focus their efforts elsewhere
- Addition of performance measurement metrics will allow us to optimize the code and determine regressions before they negatively affect users

### Next steps:

- Supporting the transition to the next version of HTR by improving packaging, technical documentation, and development utilities
- Increasing the robustness of our testing strategy
- Compare HTR's performance to a state-of-the-art solver



# Multi-physics modeling of laser-induced ignition in a multiphase rocket-module combustor

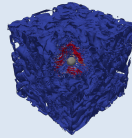
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## Sub-grid turbulence

- Large Eddy Simulation (LES) of compressible, reactive, multi-component flow
- Unresolved sub-grid scale (SGS) terms modeled with Smagorinsky model and constant turbulent Prandtl and Schmidt numbers
- Spatial discretization with hybrid SS/ENO type schemes

Laser ignition in HIT used as a unit test for SGS model and turbulent combustion model assessment and verification



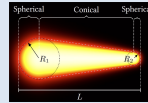
0 20 40 60 80  $\mu_{\text{sgs}}/\mu$



SGS model performs correctly: contribution of eddy viscosity decreases with increased resolution

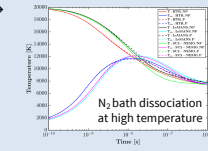
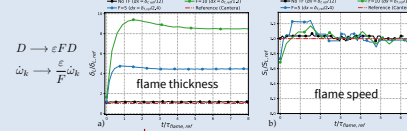
## Combustion modeling

Finite-rate chemistry (12 species, FFCM<sup>2</sup>)



Simplified kernel representation as a laser-equivalent heat source<sup>2</sup>

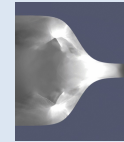
Turbulent combustion effects captured via the thickened flame treatment → laminar flame speed preserved



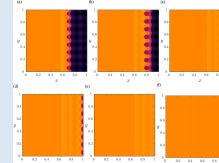
Non-equilibrium thermodynamics modeled using two-temperature models

## Outflow BC

- NSCBC extended to multi-physics problems (e.g. non-equilibrium thermodynamics)
- Substantially reduce reflections into the domain



Shock/entropy-wave interaction



## Diffuse interface modeling for two-phase flow

- 4eqn model<sup>3</sup> enforces **mechanical + thermal equilibrium** → direct integration into a multi-physics setting
- Regularizing terms in diffuse interface models maintain an interface thickness on the order of the grid-size

$$\frac{\partial m_i}{\partial t} + \nabla \cdot (\hat{m}_i \mathbf{u}) = \nabla \cdot \hat{\mathbf{R}}_i, \quad i = 1, 2$$

$$\frac{\partial \hat{m}}{\partial t} + \nabla \cdot (\hat{m} \hat{\mathbf{u}} + \hat{p} \mathbf{k}) = \nabla \cdot (\hat{\mathbf{f}} + \hat{\mathbf{a}})$$

$$\frac{\partial \hat{E}}{\partial t} + \nabla \cdot (\hat{E} \hat{\mathbf{u}} + \hat{p} \mathbf{u}) = \nabla \cdot (\hat{\mathbf{r}} + \hat{\mathbf{u}}) + \nabla \cdot (\lambda \nabla T) + \sum_{i=1}^2 \hat{f}_i + \sum_{i=1}^2 \hat{q}_i \cdot \nabla \hat{\phi}_i$$

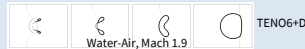
$\hat{R}_i = \rho_i (\nabla \phi_i - \phi_i (1 - \phi_i) \frac{\nabla \phi_i}{|\nabla \phi_i|})$ ,  $\hat{f} = \sum_{i=1}^2 \hat{f}_i$ ,  $\hat{\mathbf{a}} = \hat{R}_i / \rho$

diffusion + sharpening

Shock capturing schemes + regularization → **immiscible phases and better boundedness**

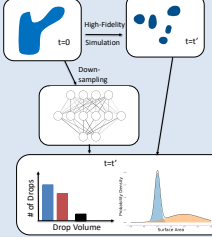


**Positivity-preserving treatment** extends the framework to handle **high-density ratios and strong shocks**



## Secondary atomization and spray

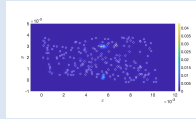
- Breakup of under-resolved drops can be modeled using **machine learning (ML)**
- Training on high-fidelity simulations of single-drop aero-breakup events
- Proof of concept studies on Taylor-Green vortices and HIT: ML model performs well in predicting (1) breakup probability (2) histogram of daughter drops (3) surface area



aero-breakup at high slip velocities leads to a very fine spray<sup>4</sup>



Very high velocities → violent breakup events → smaller daughter drops → more surface area → rapid evaporation → **particle tracking after breakup may be bypassed by assuming direct evaporation**



small errors in predicting vapor mass fraction in a simplified model problem

## Contributors

Zoe Barbeau, Chris Cundy, Mario Di Renzo, Suhas Jain, Charli e Laurent, Donatella Passiatore, Jonathan Wang, Christopher Williams

## References

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 [2] Wang, J. M., Panelli, M., & Freund, J. B. (2021). Thermal effects mediating the flow induced by laser-induced optical breakdown. *Physical Review Fluids*, 6(6), 063403.  
 [3] Jain, S. S., Adler, M. C., West, J. R., Mani, A., Moin, P., & Lele, S. K. (2023). Assessment of diffuse-interface methods for compressible multiphase fluid flows and elastic-plastic deformation in solids. *Journal of Computational Physics*, 475, 111866.  
 [4] Theofanous, T. G., & Li, G. J. (2008). On the physics of aerobreakup. *Physics of fluids*, 20(5), 052103.

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